



**Combustion Science and Technology** 

ISSN: (Print) (Online) Journal homepage: https://www.tandfonline.com/loi/gcst20

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To cite this article: Liuhao Ma, Kin-Pang Cheong, Min Yang, Chaokai Yuan & Wei Ren (2021): On the Quantification of Boundary Layer Effects on Flame Temperature Measurements Using Line-of-sight Absorption Spectroscopy, Combustion Science and Technology, DOI: 10.1080/00102202.2021.1925657

To link to this article: <u>https://doi.org/10.1080/00102202.2021.1925657</u>



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# On the Quantification of Boundary Layer Effects on Flame Temperature Measurements Using Line-of-sight Absorption Spectroscopy

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#### ABSTRACT

We studied the boundary layer effects on flame temperature measurements using line-of-sight (LOS) absorption spectroscopy. The primary objective is to quantify the measurement uncertainty of the LOSdetermined temperature, which is influenced by the thermal and species boundary layers existing in standard laminar premixed flames. The boundary layer thickness ( $\delta$ ), central flame temperature ( $T_c$ ) and species concentration  $(X_c)$  are the major factors investigated in this work. Typical absorption lines of H<sub>2</sub>O in the wavelength range of 1.4–2.9 µm were examined under different boundary layer conditions  $(\delta/L = 0-50\%, L$  is the optical path length above the flame;  $T_c = 1400-$ 2200 K; and  $X_c = 14-20\%$ ). As a result, the thermal boundary layer is observed to contribute mainly to the temperature deviation of the LOS measurement, which increases with  $\delta$  and  $T_c$ . In comparison, the variation of  $X_c$  between 14% and 20% has a negligible effect on the temperature measurement in the presence of both thermal and species boundary layers. A systematical investigation of all the selected line pairs reveals that a properly selected line pair reduces the temperature deviation by a maximum of 16.8% under typical laminar flame conditions. The particular line pair centered at 4029.52  $cm^{-1}$ and 4030.73 cm<sup>-1</sup>, which could be detected by a single tunable semiconductor laser, is recommended for the LOS temperature measurement over a temperature range of 1400-2200 K. Finally, we performed a case study of five representative temperature measurements in laminar flames and successfully corrected the LOS-determined temperature by taking into account the boundary layer effects.

#### **ARTICLE HISTORY**

Received 7 December 2020 Revised 31 March 2021 Accepted 30 April 2021

#### **KEYWORDS**

Flame temperature; Boundary layer effects; Laser absorption spectroscopy; Line-of-sight

# Introduction

Laser diagnostics have been rapidly developed and extensively used in the past decade for combustion research due to their non-intrusiveness and superior spatio-temporal resolution (Goldenstein et al. 2017). The representative laser-based techniques for combustion include Raman scattering (Cantu et al. 2018), laser-induced fluorescence (Zhou et al. 2015), absorption and dispersion spectroscopy (Ma et al. 2018a, 2019), laser-induced

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incandescence (Chatterjee and Gülder 2018), and laser-induced breakdown spectroscopy (Letty et al. 2010). Among these optical techniques, laser absorption spectroscopy (LAS) represents one of the most widely-used methods owing to its fast, sensitive and quantitative measurement with a relatively simple optical configuration (Hanson 2011; Liu and Xu 2019).

As a typical line-of-sight (LOS) measurement, LAS is readily applicable to the uniform flow field where temperature, species concentration, pressure and velocity are constant along the optical path. However, such an ideal flow does not exist in actual combustion environments due to the influence of boundary layers. When sensing reactive flows with non-negligible thermochemical gradients, the LOS measurement is potentially prone to a large measurement uncertainty (Zhou et al. 2003). Therefore, the effects of nonuniformity on LOS measurements and the methods to alleviate these effects have been explored for a long term by the combustion community.

In the previous studies, the boundary layer has been identified as a major region that affects the accuracy of LOS measurement, even when the boundary layer is very thin compared to the main flow (Liu 2006). Zhou et al. (2003) observed a temperature deviation of ~140 K for LOS flame measurements caused by the boundary layer effect. Liu et al. (2005) experimentally investigated the non-uniform temperature distribution by evaluating different absorption line pairs and discovered the hot thermal boundary layer near the combustor walls. Schoenung and Hanson (1980) proposed the concept of *effective* absorption path with the assistance of probe measurements to correct the boundary layer effect on absolute species concentration measurements in flames.

The methods of multi-line thermometry (Liu, Jeffries, Hanson 2007; Ma, Lau, Ren 2017; Sanders et al. 2001; Zhang et al. 2016) and tomographic reconstruction (Cheong et al. 2019; Liu et al. 2018; Nau et al. 2015; Wei et al. 2018; Xia et al. 2017) were often used to characterize the thermochemical distribution within the boundary layer. However, more complex signal and data processing algorithms are required in these methods. Hence, researchers also spent numerous efforts in minimizing the influence of boundary layers so that the LOS measurement is still representative for the central flame temperature. Wagner et al. (2012) explored the method of using absorption lines that are insensitive to the temperature gradient region for species concentration measurements in laminar counter-flow diffusion flames. Goldenstein et al. (2013) used two absorption transitions with linear temperature-dependent line-strengths to reduce the evident discrepancy between the path-integrated and the path-averaged absorption profiles in the presence of thermochemical boundary layers. Sepman et al. (2017) exploited transitions with the modest temperature dependence to minimize the boundary layer effects on temperature measurements in a biomass gasifier.

Although the boundary layer effect has been identified as an important factor affecting LOS measurements, very few studies have been performed to quantify the influence of boundary layers, particularly on the LOS-determined temperature and species concentration. Recently Qu, Werhahn, and Ebert (2018) quantified the thermal boundary layer effect on  $CO_2$  concentration measurements in a two-zone high-temperature gas cell by taking into account the temperature gradient effect.

As temperature is very important for understanding the physicochemical process in fundamental combustion studies, the primary objective of this work is to provide a systematic and quantitative investigation of boundary layer effects on temperature measurements in standard laminar premixed flames. The axisymmetric laminar flame (i.e., McKenna burner) normally contains a uniform hot-core region and a boundary layer with thermochemical gradient. Based on the two-line absorption thermometry of  $H_2O$ , nine commonly used line pairs in the spectral range between 1.4 µm and 2.9 µm are selected for investigation. The boundary layer effects caused by both the mere thermal boundary layer and the combined thermal and species boundary layer are investigated under different flame conditions. A semi-empirical approach to obtain a more accurate central flame temperature is also proposed in this study.

#### Spectroscopic fundamentals

Here we briefly describe the spectroscopic fundamentals of LAS. The basic equation governing the LOS absorption measurement is the Beer-Lambert relation (Hanson, Spearrin, Goldenstein 2016):

$$\left(\frac{I_t}{I_0}\right)_v = \exp(-k_v L) = \exp\left(-\sum_{0}^{L} P(x)S_i(T(x))X_{abs}(x)\phi_v dx\right)$$
(1)

where  $I_0$  and  $I_t$  are the incident and transmitted laser intensities at the optical frequency v, respectively;  $k_v$  [cm<sup>-1</sup>] is the spectral absorption coefficient contributed from all the adjacent absorption transitions, L [cm] is the total path length, and the product  $k_v \cdot L$  is known as the spectral absorbance; x is the position along the optical path;  $S_i(T)$  [cm<sup>-2</sup> atm<sup>-1</sup>] is the line-strength of the specific transition i which is only a function of gas temperature T [K],  $\varphi_v$  [cm] is the line-shape function, P [atm] is the total gas pressure,  $X_{abs}$  is concentration of the absorbing species. The temperature-dependent line-strength is given by:

$$S(T) = S(T_0) \frac{Q(T_0)}{Q(T)} \left(\frac{T_0}{T}\right) \exp\left[-\frac{hcE''}{k} \left(\frac{1}{T} - \frac{1}{T_0}\right)\right] \left[1 - \exp\left(-\frac{hcv_0}{kT}\right)\right] \left[1 - \exp\left(-\frac{hcv_0}{kT_0}\right)\right]^{-1}$$
(2)

where the partition function Q(T) and line-strength  $S(T_0)$  can be found in the HITRAN database (Gordon et al. 2017). As the line-shape function  $\varphi_v$  is normalized such that its integral over the entire frequency range is unity, the spectrally integrated absorbance  $A_i$  (also known as the integrated area) can be expressed as:

$$A_i = \int_{-\infty}^{\infty} k_v L dv = \int_{0}^{L} P(x) S_i(T(x)) X_{abs}(x) dx$$
(3)

According to the two-line thermometry, temperature can be inferred from the ratio of the integrated absorbances of two absorption lines using the following equation:

$$T = \frac{\frac{hc}{k} \left( E_2'' - E_1'' \right)}{\ln(\frac{A_1}{A_2}) + \ln(\frac{S_2(T_0)}{S_1(T_0)}) + \frac{hc}{k} \frac{(E_2'' - E_1'')}{T_0}}$$
(4)

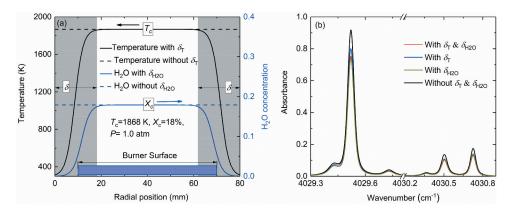
where h [J·s] is Planck's constant, c [cm·s<sup>-1</sup>] is the light speed, k [J·K<sup>-1</sup>] is the Boltzmann constant,  $v_0$  is the line-center frequency, and E' represents the lower state energy of the transition.

## Methodology

### **Boundary layer effect**

For laminar flames stabilized on a porous media or honeycomb structure, significant thermal and species concentration boundary layers exist near the flame edge. According to the previous studies (Guha and Schoegl 2014; Liu et al. 2015; Ma et al. 2020; Yu et al. 2010), both temperature and H<sub>2</sub>O concentrations along the LOS remain constant in the central region and then decreases gradually to the ambient condition. The thermal boundary layer and the H<sub>2</sub>O concentration boundary layer have almost the identical thickness. Figure 1(a) demonstrates the representative distributions of temperature and H<sub>2</sub>O concentration in a CH<sub>4</sub>/air premixed flame at the equivalence ratio  $\Phi = 0.95$  and pressure P = 1.0 atm. The central porous media of the burner has a diameter of 60 mm indicated by the blue shaded area in Figure 1(a). The gray shaded regions indicate the boundary layers. The thermal boundary layer thickness,  $\delta_T$ , is defined as the distance from the location at room temperature to that at 99% of the central flame temperature  $(T_c)$ . The species boundary layer thickness,  $\delta_{H2O}$ , is defined in a similar way.

Based on the boundary layer conditions shown in Figure 1(a), the simulated absorption spectra of H<sub>2</sub>O near 2.5  $\mu$ m are illustrated in Figure 1(b). The detailed values of the simulated peak absorbance and integrated area under different boundary layer conditions are summarized in Table S1 of the Supplementary Material. The presence of boundary layers affects the peak absorbances of both lines (4029.52 cm<sup>-1</sup> and 4030.73 cm<sup>-1</sup>) particularly for the line centered at 4029.52 cm<sup>-1</sup> due to its smaller *E*". For instance, the absorption line at 4029.52 cm<sup>-1</sup> with a smaller *E*" is more sensitive to the cold boundary layer and thus its absorbance is affected more evidently by the temperature variation along the optical



**Figure 1.** (a) Representative radial distributions of temperature and H<sub>2</sub>O concentration (gray shaded area, boundary layer; blue shaded area, central porous media; dashed line, uniform flame condition). (b) Simulated absorption lines of H<sub>2</sub>O at 4029.52 and 4030.73 cm<sup>-1</sup> under different boundary layer conditions ( $\delta_{T/L}$  and  $\delta_{H2O}/L$  are both set to be 45.4%).

path. As temperature is inferred from the ratio of the integrated absorbances shown in Eqn. (4), it is interesting to observe that the ratio of the two integrated areas remains unchanged when there is only a concentration boundary layer. Hence, the mere concentration boundary layer has no influence on the temperature measurement. Therefore, in this study we only explore the effects of the thermal boundary layer and the combined thermal/concentration boundary layer.

#### Line-pair selection

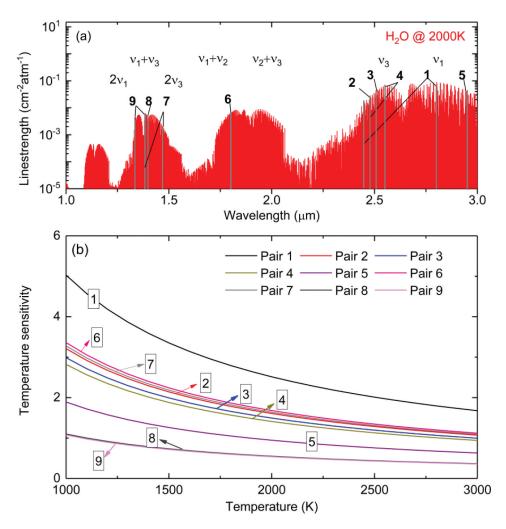
The line-selection criteria for two-line thermometry have been well documented elsewhere (Liu 2006; Zhou et al. 2003). In general, the selected absorption lines should have a good spectral isolation from nearby transitions and interfering species, a large absorption line-strength and a high temperature sensitivity. Several additional selection rules have also been proposed to reduce the boundary layer effect and improve the precision (Goldenstein et al. 2013).

Here we study all the line pairs of H<sub>2</sub>O that have been used for temperature sensing in combustion environments (Farooq, Jeffries, Hanson 2008; Goldenstein et al. 2013, 2015; Lee et al. 2020; Li, Farooq, Hanson 2011; Liu et al. 2015; Ma et al. 2020, 2018b; Peng et al. 2019, 2016; Peng, Strand, Hanson 2020; Qu et al. 2015; Wang et al. 2020, 2015; Wu et al. 2017; Zeng et al. 2018; Zhou et al. 2003). These line pairs cover the wavelength range from 1 µm to 3 µm as illustrated in Figure 2(a), belonging to the overtone  $(2v_3)$ , combination  $(v_1+v_3, v_2)$  $+v_3$ ), and fundamental ( $v_3$ ) bands. The spectroscopic parameters of these H<sub>2</sub>O lines are listed in Table 1; and these parameters are taken either from the HITRAN database (Gordon et al. 2017) or from the validation experiments (Farooq, Jeffries, Hanson 2008; Goldenstein 2014; Li, Farooq, Hanson 2011; Liu 2006; Zhou 2005). Among these H<sub>2</sub>O lines, the line pairs numbering 6-9 at 1.4  $\mu$ m and 1.8  $\mu$ m (2 $\nu_3$ ,  $\nu_1+\nu_3$ ,  $\nu_2+\nu_3$ ) have been used for flame and engine measurements by employing telecommunication fiber-coupled diode lasers (Goldenstein et al. 2015; Lee et al. 2020; Peng, Strand, Hanson 2020; Qu et al. 2015; Wang et al. 2020, 2015; Wu et al. 2017; Zeng et al. 2018). The other line pairs numbering 1–5 at 2.5–2.9  $\mu$ m ( $v_3$ ) are receiving more attention recently due to their stronger linestrengths and have been used for laboratory-scale flames (Farooq, Jeffries, Hanson 2008; Li, Farooq, Hanson 2011; Ma et al. 2018b; Peng et al. 2016) and practical propulsion engines (Peng et al. 2019). Figure 2(b) compares the temperature sensitivity of these line pairs. Although line pairs 5, 8, 9 have a low temperature sensitivity at 2000-3000 K, we still involve them for investigation considering their frequent applications in combustion diagnostics. The temperature sensitivity is defined as the derivative of Equation (4) with respect to temperature:

$$\left|\frac{dR/R}{dT/T}\right| \approx \left(\frac{hc}{k}\right) \frac{\left|E_2'' - E_1''\right|}{T}$$
(5)

### Simulation of temperature deviation

We aim to quantify the deviation of LOS-determined temperature measured under different boundary layer conditions, which are specified by the boundary layer thickness ( $\delta$ ), the central flame temperature ( $T_c$ ), and the species concentration ( $X_c$ ). The absolute



**Figure 2.** (a) Representative line pairs of  $H_2O$  previously used for temperature measurements. (b) Comparison of temperature sensitivity for the selected line pairs at 1000–3000 K.

temperature deviation ( $\Delta T = T_c - T_{LOS}$ ) is defined as the difference between the LOSdetermined temperature ( $T_{LOS}$ ) and the actual central temperature  $T_c$ . Additionally, we define the relative temperature deviation ( $\varepsilon$ ) as the ratio of  $\Delta T$  and  $T_c$ .

Figure 3 presents the flowchart of numerically evaluating the temperature deviation caused by the boundary layer effects. The input parameters include those describing the thermochemical and boundary layer conditions (e.g.,  $T_{\circ}$ ,  $T_{edge}$ ,  $X_{\circ}$ ,  $X_{edge}$ ,  $\delta$ ,  $k_{slope}$ ), the laser wavelength range covering the target spectral features, and the spectroscopic parameters. The LOS absorption spectra of the selected absorption transitions are simulated under different flame conditions. The integrated absorbance is obtained by fitting the absorption profiles using the multi-Voigt function and Levenberg-Marquardt (L-M) algorithm. The temperature deviation is ultimately determined by the difference of the derived temperature  $T_{LOS}$  and the central temperature  $T_c$ .

Line pair #	Frequency (cm <sup>-1</sup> )	Wavelength (nm)	S @296 K (cm <sup>-2</sup> atm <sup>-1</sup> )	<i>E"</i> (cm <sup>-1</sup> )	<i>∆E″</i> (cm <sup>−1</sup> )
1 <sup>a</sup>	4083.89	2448.65	$3.00 \times 10^{-8}$	4331.07	3488.71
	3565.67	2804.52	$3.36 \times 10^{-1}$	842.36	5400.74
2 <sup>b</sup>	4030.73	2480.94	$2.68 \times 10^{-9}$	4889.49	2228.54
-	4029.52	2481.69	$1.10 \times 10^{-4}$	2660.95	
3 <sup>b</sup>	3982.75	2510.83	$5.92 \times 10^{-7}$	3654.04	2072.71
	3982.06	2511.26	$9.13 \times 10^{-3}$	1581.33	
4 <sup>b</sup>	4029.52	2481.69	$1.10 \times 10^{-4}$	2660.95	1956.75
	3920.08	2550.97	$6.35 \times 10^{-1}$	704.20	
5 <sup>b</sup>	3459.73	2890.41	$5.14 \times 10^{-6}$	3386.00	1312.50
	3460.59	2889.68	$3.32 \times 10^{-3}$	2073.50	
6 <sup>b</sup>	5553.86	1800.57	$7.55 \times 10^{-7}$	3314.86	2331.95
	5554.18	1800.45	$7.80 \times 10^{-3}$	982.91	
7 <sup>b</sup>	6806.03	1469.29	$6.40 \times 10^{-7}$	3291.15	2246.09
	7185.60	1391.67	$1.96 \times 10^{-2}$	1045.06	
8 <sup>a</sup>	7153.74	1397.87	$8.59  imes 10^{-6}$	2552.86	763.82
	7154.35	1397.75	$3.68 \times 10^{-4}$	1789.04	
9 <sup>b</sup>	7444.35/7444.37	1343.30	$1.12 \times 10^{-3}$	1774.75/1806.67	745.65
	7185.60	1391.67	$1.96 \times 10^{-2}$	1045.06	

Table 1. Spectroscopic parameters of the representative line pairs.

<sup>a</sup>HITRAN 2016 database (Gordon et al. 2017);

<sup>b</sup>Validated in experiments (Farooq, Jeffries, Hanson 2008; Goldenstein 2014; Li, Farooq, Hanson 2011; Liu 2006; Zhou 2005).

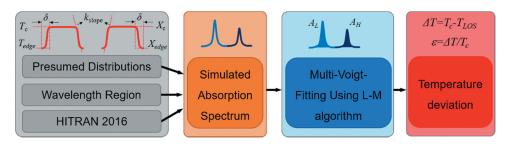


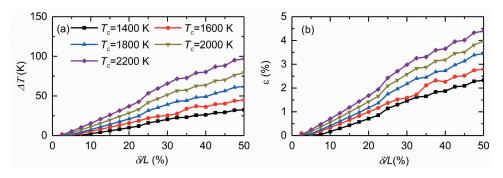
Figure 3. Flowchart of numerically evaluating the boundary layer effect on LOS-determined temperature.

To simulate the non-uniform environment, the optical path is normally discretized into numerous elements, each of which is assumed to have a uniform thermochemical property. Here an elemental length of 0.1 mm is used in the simulation for the total optical path length of 80 mm. The finer discretization has a negligible influence on the calculation result, which can be seen from Fig. S1 of the Supplementary Material. The central flame temperature investigated in this study ranges from 1400 K to 2200 K, which covers most laminar flame experiments. The representative results of the simulated absorption spectra along with the Voigt-fitting profiles are provided in Fig. S2 of the Supplementary Material.

### **Results and discussion**

#### Effect of thermal boundary layer

As the integrated absorbance is mainly dependent on temperature, the effect of the thermal boundary layer is firstly explored before considering both thermal and concentration boundary layers to be discussed in Section 4.2. In this section we discuss the variation of temperature deviation ( $\Delta T$ ) with the thickness of thermal boundary layer ( $\delta_T$ ) and the



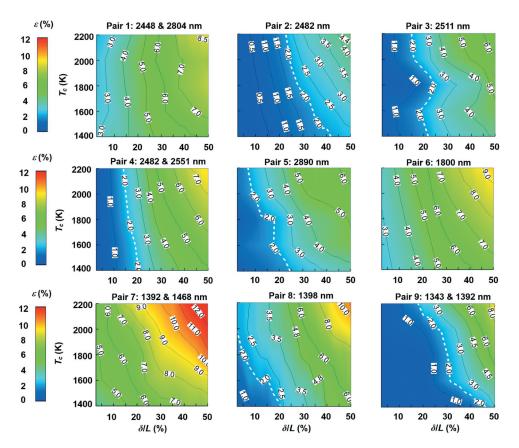
**Figure 4.** Variation of LOS-determined temperature with boundary layer thickness for different values of  $T_c$ : (a) absolute deviation ( $\Delta T$ ), (b) relative deviation ( $\epsilon$ ).

central flame temperature  $(T_c)$ . First, we take the line pair 2 with the largest E" as an example to interpret the simulation results. Figure 4 plots the variation of the absolute  $(\Delta T)$  and relative  $(\varepsilon)$  deviations of temperature with the boundary layer thickness. It is evident that the LOS-determined temperature deviates further from  $T_c$  with the increased  $\delta$ . A deviation of ~ 100 K (or ~4.4%) is observed at the highest temperature of 2200 K. At the fixed  $\delta$ , the deviation (either  $\Delta T$  or  $\varepsilon$ ) increases with the central flame temperature  $T_c$ .

To better view the thermal boundary effect, we summarize the results of all the selected line pairs in a contour plot illustrated in Figure 5. The deviation always has a positive value for all the line pairs, indicating that the LOS-determined temperature is consistently lower than  $T_c$ . At an arbitrary  $T_c$  between 1400 K and 2200 K, the deviation increases with the boundary layer thickness. Due to the different sensitivities to the thermal boundary layer, all these selected line pairs demonstrate very different results of temperature deviations. For instance, line pairs 7 and 8 have the largest deviation over 10% as shown in Figure 5; but line pairs 2, 3 and 5 show a deviation mostly below 6% even for a very large boundary layer thickness ( $\delta/L = 50\%$ ). This is mainly because these three line pairs have a larger E'' ( $E_s'' >$ 1581.33 cm<sup>-1</sup>,  $E_L'' >$  3386.00 cm<sup>-1</sup>). Particularly, line pair 2 shows the best immunity to the effect of thermal boundary layer over the entire temperature range and under different boundary layer conditions.

It is of interest to find out the boundary layer condition, corresponding to a very small temperature deviation, so that the LOS measurement is still valid. Here we define a critical boundary layer thickness ( $\delta_{critical}$ ), which leads to a  $\varepsilon$ -value of 2% or less. Such a  $\delta_{critical}$  at different temperatures could be readily identified in Figure 5 for line pairs 2–5, 8 and 9. Hence, the accuracy of the LOS temperature measurement is acceptable when the critical condition of boundary layer satisfies. A good example is the temperature measurement above a long flat burner (Farooq, Jeffries, Hanson 2008), which has a narrow non-uniform flame region ( $\delta/L < 4\%$ ). The LOS-determined temperature using line pair 3 was in an excellent agreement with the thermocouple result (Farooq, Jeffries, Hanson 2008).

It is known that the boundary layer mainly influences the integrated area of absorbance due to the varied line-strength along the absorption path. Here we take line pair 4 as an example to quantify the effect of boundary layer thickness on the temperature deviation. Figure 6(a) plots the temperature-dependent line-strengths of the two absorption lines centered at 4029.52 cm<sup>-1</sup> and 3920.08 cm<sup>-1</sup>, respectively. The corresponding line-strength

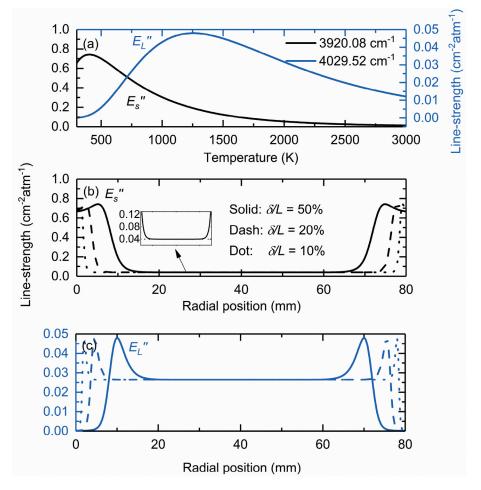


**Figure 5.** Contour of the temperature deviation ( $\epsilon$ ) caused by the thermal boundary layer for all the 9 line pairs listed in Table 1.

distributions along the absorption path are plotted in Figure 6(b,c), respectively, for a central flame temperature of 2200 K and ambient temperature of 300 K. Three different boundary layer conditions ( $\delta/L = 10\%$ , 20% and 50%) are investigated for comparison. Note that the total optical path (2 L) is fixed while the boundary layer thickness (2 $\delta$ ) varies. Figure 6(b) shows that the absorption line with a smaller  $E^{"}$  (denoted as  $E_s"$ ) has a larger linestrength within the boundary layer than that in the central flame, leading to a larger integrated area of absorbance  $A_s$ . In comparison, the other line with a larger E" (denoted as  $E_L"$ ) undergoes an initial increase of the line-strength followed by a decline from the central flame to the edge. As a result, an overall decrease in the integrated area  $A_L$  of absorbance is obtained as illustrated in Figure 6(c). Hence, the ratio of these two integrated areas increases due to the existence of the thermal boundary layer, which under-estimates the central flame temperature. A similar line-strength plot for line pair 2 is illustrated in Fig. S3 of the Supplementary Material.

#### Combined thermal and concentration boundary layer

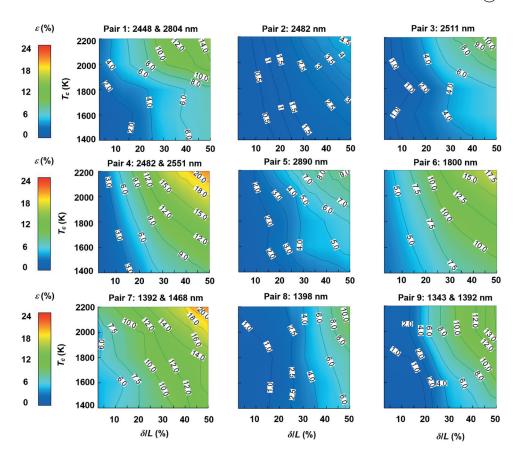
The concentration boundary layer always accompanies with the thermal boundary layer. Hence, the combined effects of thermal and concentration boundary layers are investigated



**Figure 6.** (a) Temperature-dependent line-strengths of the two absorption lines in line pair 4. (b) Radial distribution of line-strength of the absorption line with a smaller  $E_s''$ . (c) Radial distribution of line-strength of the absorption line with a larger  $E_L''$ . Flame condition:  $T_c = 2200$  K;  $\delta/L = 10\%$ , 20% and 50%, respectively.

in terms of  $\delta$ ,  $T_c$  and  $X_c$ . As discussed in Section 3.1, the thickness of the concentration boundary layer is treated the same as that of the thermal boundary layer.

Figure 7 illustrates the contour plots of temperature deviation for all the line pairs by varying  $\delta$  and  $T_c$ . The temperature deviation increases gradually with  $\delta$  over the entire temperature range. The largest deviation appears at the top-right corner of the contour plot for each line pair, corresponding to the flame with the highest central temperature and the largest boundary layer thickness. In particular, the maximum  $\varepsilon$  ranges from 4.8% for line pair 2 to 21.6% for pair 7 under the condition of  $\delta/L = 50\%$  and  $T_c = 2200$  K. By comparing the results to Figure 5, the additional concentration boundary layer increases the temperature deviation of most line pairs except the line pairs 2 and 8. It is of interest to observe that the combined thermal/concentration boundary layers for line pairs 2 and 8 have almost the identical effect to that caused by the mere thermal boundary layer. These two line-pairs have



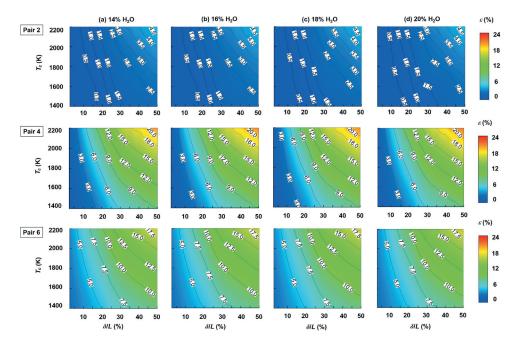
**Figure 7.** Contour of temperature deviation ( $\epsilon$ ) caused by the combined temperature and concentration boundary layers for all the 9 line pairs.

much smaller line-strengths at low temperatures, making them insensitive to the concentration variation in the cold boundary layer.

The variation of the central gas composition may also affect the inferred temperature, which is investigated under different flame conditions. Figure 8 presents the simulation results of three line pairs 2, 4, and 6 as an example. For the typical CH<sub>4</sub>/air laminar premixed flames with  $\Phi = 0.7$ –1.2, the H<sub>2</sub>O concentration is known to be in the range of 15%–19%; and the H<sub>2</sub>O concentration is much lower for the other hydrocarbon/air (e.g., C<sub>2</sub> H<sub>4</sub>, C<sub>3</sub>H<sub>8</sub>) flames or the flames with N<sub>2</sub>/CO<sub>2</sub> dilutions. Therefore, the H<sub>2</sub>O concentration is varied over a wide range from 14% to 20% for investigation. Although the integrated absorbance of the individual absorption line varies with the gas composition, the calculated temperature deviation is almost irrelevant to the variation of the gas composition as illustrated in Figure 8.

#### Case study in flame measurement

Our numerical analysis reveals that the LOS-determined temperature is lower than the actual temperature of the central flame by an amount of  $\Delta T$ . Here case studies of the



**Figure 8.** Contour of temperature deviation caused by the combined temperature and concentration boundary layers for line pairs 2, 4 and 6. The  $H_2O$  concentration at the central flame is varied between 14% and 20% for investigation.

Table 2. Experimental details of the five representative laser absorption measurements.

	Case 1 <sup>This work</sup>	Case 2 This work	Case 3 <sup>(Zhou et al. 2003)</sup>	Case 4 <sup>(Fu et al. 2020)</sup>	Case 5 This work
Line pair	2	4	6	8	9
Reactants	CH₄/air	CH₄/air	$C_2H_4/air$	CH₄/air	CH₄/air
Equivalence ratio	0.8-1.2	0.8-1.2	0.65	0.85-1.22	0.8-1.2
Burner type	McKenna	McKenna	Hencken	McKenna	McKenna
Co-flow	Nitrogen	Nitrogen	NA	NA	Nitrogen
HAB	5 mm	5 mm	10 mm	6 mm	5 mm

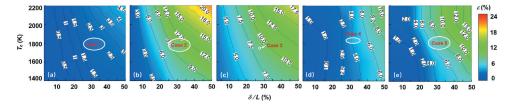
previous LOS flame measurements are performed based on the conclusions made in this work.

The experimental conditions are summarized in Table 2. For the McKenna burner, measurements were performed at the height above the burner (HAB) of 5–6 mm. For the Hencken burner, measurements were taken at the HAB of 10 mm. For the CH<sub>4</sub>/air flames studied in cases 1, 2 and 5, the equivalence ratio was varied between 0.8 and 1.2 by changing the air flow rate but fixing the CH<sub>4</sub> flow rate at 1.5 L/min. The optical setup is almost identical to the recent study described in (Ma et al., 2018b), which is also provided in Fig. S4 of the Supplementary Material. The reference central flame temperature was obtained using the thermocouple, the LAS profile-fitting method, and the CFD simulation (Ma et al. 2020). For case 3, the flow rates of C<sub>2</sub>H<sub>4</sub> and air were set to be 2.9 L/min and 64 L/min, respectively, to obtain an equivalence ratio of 0.65 (Zhou et al. 2003); and the central flame temperature

was measured by the thermocouple and the profile-fitting method (Zhou et al. 2003). For case 4, the varied equivalence ratios ( $\Phi = 0.85$ –1.22) were obtained by changing the CH<sub>4</sub> flow rate while fixing the air flow rate at 14.5 L/min (Fu et al. 2020). The CFD simulation was performed using the recent numerical framework (Ma et al. 2020) and the simulated central flame temperature was used as the reference. With the knowledge of boundary layer thickness and the reference central temperature, the temperature deviations can be determined. In general, such information can be obtained using an easily accessible thermocouple, a low-cost infrared CCD camera (pyrometry), or even the CFD simulation. For the laminar flame investigated in the current work, we used the results from the previous experiments (e.g., thermocouple measurement) and the CFD simulations (Fu et al. 2020; Ma et al. 2020; Zhou et al. 2003). The flame conditions for the above cases are added to the corresponding temperature deviation contours shown in Figure 9. For each of the case listed in Table 2, the temperature deviations of all the measurements fall into the shaded region in Figure 9. Although cases 1 and 4 have a relatively small deviation (1.5-2.7% and 2.7-5%, respectively), the other cases (cases 2, 3, and 5) have a very large deviation of 6-12% which could not be accepted for temperature determination.

Figure 10 presents the original LOS-determined temperature without correction, the reference  $T_c$ , and the corrected LOS-determined temperature for each case. The overall temperature can be obtained by adding the correction determined from Figure 9 to the original LOS measurement. As illustrated in Figure 10(a), the original LOS-determined temperature was consistently lower than the reference  $T_c$ . For cases 1, 2 and 5 (equivalence ratio  $\Phi = 1.0$ ), the boundary layer thickness was evaluated to be 12 mm ( $\delta/L = 30\%$ ) based on the experimental measurement using the thermocouple measurement and CFD simulation. The relative temperature deviations were determined to be 1.7% (case 1), 12% (case 2) and 7.8% (case 5), respectively, according to the boundary layer effects shown in Figure 9. The overall corrected LOS-determined temperature is in good agreement with the reference values mostly within a difference of 1.2%; the maximum difference between the corrected LOS-determined temperature is 3.6% at  $\Phi = 0.8$  for case 5.

For case study 3, with the temperature deviation of 155 K considered as shown in Fig. 11, the LOS-determined temperature was corrected to be 1775 K, showing a good agreement with the thermocouple measurement (1740 K) and the LAS profile-fitting result (1760 K). Similarly, the LOS-determined temperature for case 4 is always below the reference temperature when  $\Phi$  is changed from 0.85 to 1.23. The corrected LOS-determined temperature agrees well with the CFD simulation within a relative difference of 1.0%.



**Figure 9.** Temperature deviations of the investigated cases: (a) case 1, (b) case 2, (c) case 3, (d) case 4, and (e) case 5.

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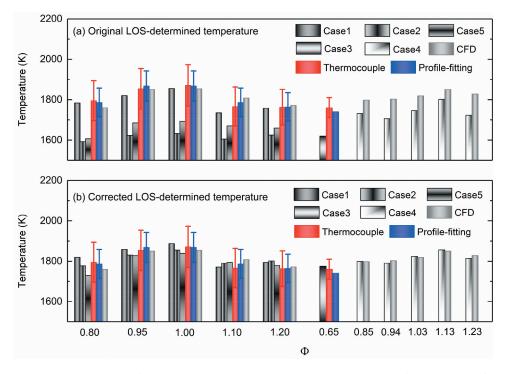


Figure 10. (a) Comparison of the original LOS-determined temperature and the reference central flame temperature. (b) Comparison of the corrected LOS-determined temperature and the reference central flame temperature for the five representative cases listed in Table 2.

### Conclusions

We performed a quantitative investigation of the boundary layer effects on temperature measurements using LOS absorption spectroscopy in laminar premixed flames. The temperature deviation between the LOS-determined temperature and the actual central flame temperature was evaluated by considering the thermal and concentration boundary layers. As a demonstration, we studied a total of nine line pairs of H<sub>2</sub>O that have been widely used for temperature sensing in combustion environments. Compared with the species concentration boundary layer, the thermal boundary layer has a more significant influence on the LOSdetermined temperature. The temperature deviation increases gradually with  $\delta$  and  $T_c$  over the entire temperature range of 1400–2200 K. The additional species concentration boundary layer increases the temperature deviation for most line pairs. However, the temperature deviations are irrelevant to the variation of  $X_c$  between 14% and 20% in the presence of both thermal and species concentration boundary layers. Among all the line pairs investigated in this study, the largest temperature deviation of 21.6% is expected for a boundary layer thickness of  $\delta/L$  = 50% and central flame temperature of 2200 K. The line pair of 4029.52 cm<sup>-1</sup> and 4030.73 cm<sup>-1</sup> is recommended for the LOS temperature measurement at 1400–2200 K due to its good immunity to boundary layer effects. As a general rule to further reduce the measurement uncertainty, we suggest the selection of line pairs with larger E, larger difference of E, smaller line-strength at low temperature and larger line-strength at high temperature. By performing the case studies of flame measurements, the LOS-

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determined temperature could be corrected by taking into account the boundary layer effects. We expect this work to be valuable for a better design of two-line thermometry and a more accurate interpretation of the experimental data obtained in flame studies.

# Nomenclature

$k_v$ Gas absorption coefficient at frequency $v$ (cm <sup>-1</sup> ) $l_0$ Incident light intensity $l_t$ Transmitted light intensity $S_i(T)$ Line-strength (cm <sup>-2</sup> atm <sup>-1</sup> ) of the specific transition $i$ $S_i(T_0)$ Line-strength at a reference temperature $T_0$ (296 K) $\varphi_v$ line-shape function (cm)
$I_t$ Transmitted light intensity $S_i(T)$ Line-strength (cm <sup>-2</sup> atm <sup>-1</sup> ) of the specific transition i $S_i(T_0)$ Line-strength at a reference temperature $T_0$ (296 K)
$S_i(T)$ Line-strength (cm <sup>-2</sup> atm <sup>-1</sup> ) of the specific transition i $S_i(T_0)$ Line-strength at a reference temperature $T_0$ (296 K)
$S_i(T_0)$ Line-strength at a reference temperature $T_0$ (296 K)
P Total gas pressure (atm)
X <sub>abs</sub> Absorbing gas species concentration
Optical path length (cm)
A <sub>i</sub> Integrated absorbance (cm <sup>-1</sup> ) of the specific transition i
h Planck's constant (J s)
c Light speed (cm/s)
k Boltzmann constant (J/K)
<i>E</i> " Lower state energy (cm <sup>-1</sup> )
$\Delta E''$ Difference of lower state energy (cm <sup>-1</sup> )
Q(7) Partition function at temperature T
$Q(T_o)$ Partition function at a reference temperature $T_o$ (296 K)
T <sub>c</sub> Central flame temperature (K)
<i>X<sub>c</sub></i> Central species concentration
ΔT Absolute temperature deviation (K)
ε Relative temperature deviation (%)
δ Boundary layer thickness
$\delta_{critical}$ Critical boundary layer thickness
$\delta_{\tau}$ Thermal boundary layer thickness
$\delta_{H2O}$ Boundary layer thickness of H <sub>2</sub> O concentration profile
<i>T<sub>edge</sub></i> Temperature near the flame edge
X <sub>edge</sub> Species concentration near the flame edge

#### Funding

This research is supported by National Natural Science Foundation of China (NSFC) (51776179), Natural Science Foundation of Guangdong Province (2019A1515011372), Science Foundation of Sichuan Province (No. 2020JDRC0034), and Open Funding from State Key Laboratory of Hightemperature Gas Dynamics . We thank Professor Xing Chao and Mr. Zhenhai Wang from Tsinghua University for providing the set of flame temperature data.

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